## PHYS 7810 Hydrodynamics Spring 2024

## Lecture 14 Boundary layers

February 29

Navier-Stokes equations: incompressible: 
$$\nabla \cdot \vec{v} = 0$$

Set  $t \cdot \nabla \vec{v} + \frac{\partial \vec{v}}{\partial t} = \nu \nabla^2 \vec{v}$ 

static (assumption)

Today: flow around thin plate... assume no-slip BC:

 $\vec{v} = \vec{0}$ 

Take  $\vec{v} = v_0 \hat{x}$  far from plate

Physical picture: flow is  $\vec{v}$  const. Thomogeness... outside of bdy layer

boundary layer

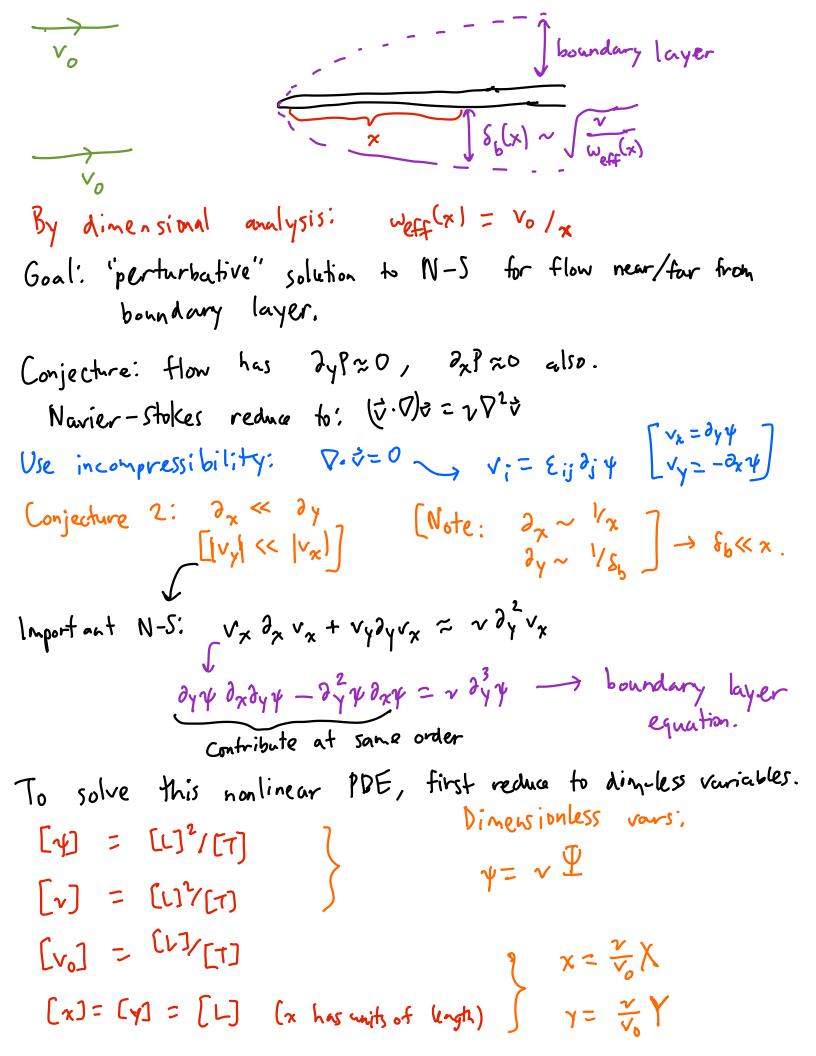
Today:  $\vec{v} = \vec{v} = \vec{v}$ 

Warm-up: infinite plate  $\Rightarrow V_{x}(y\rightarrow a, t)=f(t)$ Y=0 Assume: translation invariance in x (0x=0) incompressible: 2xxx + dyry =0 Thus:  $v_y = g(x, t) = 0$  by looking at y = 0.

avier - Stokes:

Thus:  $v_y = g(x, t) = 0$  by looking at y = 0.

Further forms x-Navier-Stokes:  $g_{t} \wedge x + \frac{\lambda^{2} \sigma^{2} \sigma^{2}}{\sigma^{2}} + \frac{\lambda^{2} \sigma^{2} \sigma^{2}}{\sigma^{2}} = \lambda^{2} \sigma^{2} \wedge x + \sigma^{2} \tau(f)$ Take t-Fourier transform: v(t) -> v\_x(w) -iwîn= v dyîn -iwflw)
4 Solved by în(y,w)=f(w) + Ae+Jiy,y+Be-Jiwy  $\int_{-i}^{-i} = \frac{1-i}{\sqrt{2}} = e^{-i\pi/4}$ A=0 (regularity at  $y=\infty$ )  $B=-\hat{f}(\omega)$  due to no-slip. Thus: \$(y, w) = f(w)(1-e= 1/2) = e-1/26 (Ng: w>0) here i Sb = boundary layer thickness = Jw Outside y>> 8, v~v~ (irrotational solla, neglecting In our original problem, no explicit t-dep.



We're going to solve for 
$$\Psi(x,y)$$
:  $v_x = v_0 \partial_y \Psi$ 
 $v_y = -v_0 \partial_x \Psi$ 

and:  $\partial_y \Psi \partial_x \partial_y \Psi - \partial_x \Psi \partial_y^2 \Psi = \partial_y^2 \Psi$ 

Conjecture 3: Similarity ansatz:

 $\Psi = \chi^{\alpha} f(\frac{y}{s_0})$ . For now, when are undetermined

befine as  $g = \chi^{\alpha} f(\frac{y}{s_0})$  befine as  $g = \chi^{\alpha} f(\frac{y}{s_0})$ .

 $\partial_y \Psi = \chi^{\alpha-2} f(\frac{y}{s_0}) - \beta \chi^{\alpha} \chi^{\alpha} f'(\frac{y}{s_0}) = \chi^{\alpha-2} f''(\frac{y}{s_0})$ 

only compatible  $\psi$  sin:  $\chi = -3\beta = 2\alpha - 2\beta - 1$ 

Fix  $g = \chi^{\alpha} f(\frac{y}{s_0}) - \chi^{\alpha} f(\frac{y}{s_0}) = \chi^{\alpha-2} f(\frac{y}{s_0})$ 
 $\partial_y \Psi |_{Y=\infty} = \frac{1}{2} = \chi^{\alpha-\beta} f'(s_0)$ 

Solve for  $f$ :

 $2^{\alpha} \Psi = \chi^{\alpha} f''(\frac{y}{s_0}) = \chi^{\alpha-\beta} f'(\frac{y}{s_0}) + \frac{1}{2} \chi^{\alpha-\beta-1} f' - \beta_x \chi^{\alpha-\beta-1} f''$ 
 $\chi^{\alpha} = \chi^{\alpha-\beta} f''(\frac{y}{s_0}) = \chi^{\alpha-\beta} f'(\frac{y}{s_0}) + \frac{1}{2} \chi^{\alpha-\beta-1} f'' - \beta_x \chi^{\alpha-\beta-1} f''$ 

Deduce:  $\chi^{\alpha} = \chi^{\alpha-\beta} f''(\frac{y}{s_0}) = \chi^{\alpha-\beta} f''(\frac{y}{s_0}) + \frac{1}{2} \chi^{\alpha-\beta-1} f'' + \frac{1}{2} \chi^{\alpha-\beta-$ 

Numerically solve! boundary layer fins fresh fr When are approximations accurate?  $\frac{\partial y \Psi}{\partial x} \gg \frac{\partial x \Psi}{\partial x}$  or:  $f'(S) \gg \frac{1}{2\sqrt{x}} [f - Sf']$ This is accurate if  $X \gg 1$ , or  $X = x \stackrel{\vee}{\sim} \gg 1$  Then  $x^2 \gg \frac{v\pi}{v_0} = S_h^2$ , or  $x \gg \delta_b$ Approx fails close to onset of boundary layer at x=0. Need to check that pressure gradients are small. Already used x-NS to find solla... Check approx by looking at y-NS; using method of Anylow = de + Anxlow + Anylow dominant balance, ν<sub>ο</sub>, ν<sub>γ</sub> ~  $\frac{\nu}{5^{2}_{b}}$  ν<sub>γ</sub>  $\frac{\Delta l}{l}$  across boundary layer  $\sim \left[\frac{\nu}{s_b^2}v_y\right] \cdot s_b \sim \frac{\delta_b}{\chi}v_o v_y \sim \frac{\delta_b}{\chi}v_o \cdot v_o \frac{\delta_b}{\chi}$ So  $\Delta P \sim \left(\frac{\delta b}{\lambda}\right)^2 v_0^2$  and  $\frac{\partial x P}{P} \sim \frac{v_0^2}{\lambda} \left(\frac{\delta b}{\lambda}\right)^2 \sim \left(v_{\star}^2 \lambda v_{\lambda}\right) \cdot \left(\frac{\delta b}{\lambda}\right)^2$ leading order Shall term in a-NS Correction!

What is the drag force on plate (per unit width!)

$$F_{\gamma} = 2\int dx \left(-\tau_{\gamma\gamma}\right) = 2\eta \int dx \, \partial_{\gamma}v_{\chi} = 2\eta \int dx \, \partial_{\gamma}^{2} \psi$$
type below

$$= 2\eta \frac{v_{0}^{2}}{v} \int_{0}^{1} A\left(\frac{v_{0}}{v_{0}}\right) \frac{f''(s)}{JX} \int_{0}^{1} s ds$$

$$= 2\eta v_{0} \int_{0}^{1} A\left(\frac{v_{0}}{v_{0}}\right) \frac{f''(s)}{JX} \int_{0}^{1} s ds$$

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$$= 2\eta v_{0} \int_{0}^{1} A\left(\frac{v_{0}}{v_{0}}\right) \frac{f''(s)}{JX} \int_{0}^{1} s ds$$

$$= 4f''(s) \cdot p \int_{0}^{1} v_{0}^{3} L$$

Our calculation valid if Ly  $\int_{0}^{1} v_{0}^{3} L$ 

windth

for approx boundary (ager flow.)

Fixth = var. 4 p v\_{0}^{2} L f''(s) \cdot R^{-1/2} = p v\_{0}^{3} A \cdot C\_{0}

while:

$$C_{0} \approx 4 f''(s) R^{-1/2} \approx 1.3 R^{-1/2}.$$

Compare to creep flow (lec 13):

$$F_{\chi, tot} = 6\pi p v R v_{0} = \frac{p v_{0}^{2}}{2} A \cdot \frac{3}{R} \quad creep flow.$$

Compare to creep flow (lec 13):

$$C_{0} \approx R^{-1} \int_{0}^{1} R \int_{0}^{1} r r^{2} r^{2$$